A higher-order space-time discontinuous Galerkin method is presented for the simulation of compressible flows. The effect of the discrete formulation on the nonlinear stability of the scheme is assessed through numerical simulations. For marginally resolved turbulent simulations at moderate Reynolds number, polynomial dealiasing is shown to be necessary in order to maintain stability at high order. With increasing Reynolds number, the formulation using conservative variables is shown to be unstable at high order even when using polynomial dealiasing. Using an entropy variable formulation consistent with established entropy stability theory ensures nonlinear stability at high and infinite Reynolds number. The effect of the numerical flux for underresolved turbulent simulations is investigated. A low-Mach modified flux term is presented to suppress the biased pressure-dilatation term seen with other upwind numerical fluxes. Subgrid-scale modeling effects of different numerical flux functions on the kinetic energy spectrum are examined in the limit of infinite Reynolds number.

I. Introduction

Turbulent flows have a large range of spatial and temporal scales which need to be resolved in order to obtain accurate predictions. Higher-order methods can provide greater efficiency for simulations requiring high spatial and temporal resolution, allowing for solutions with fewer degrees of freedom and lower computational cost than traditional second-order computational fluid dynamics (CFD) methods.\(^1\) Higher-order methods have been widely used for turbulent flows. However, the reduced numerical stabilization present in higher-order schemes implies that special care needs to be taken in the development of numerical methods to suppress nonlinear instabilities.\(^2\)\(^-\)\(^6\) In this work we present the development of a higher-order space-time discontinuous Galerkin method with a focus on the aspects of our numerical scheme required for ensuring nonlinear stability for turbulent simulations at high Reynolds numbers.

The importance of nonlinear stability was recognized in the first direct numerical simulation (DNS) of incompressible isotropic turbulence using spectral methods.\(^2\) The quadratic nonlinearities in the inviscid fluxes lead to aliasing errors when using a collocated or pseudo-spectral scheme. Over long-time evolution, these aliasing errors can lead to non-physical accumulation of energy and ultimately catastrophic failure of the numerical simulation. Evaluating the fluxes in the physical space using a dealiasing rule with 3/2 N points results in a Galerkin formulation (i.e., exact projection of the fluxes) and eliminates the aliasing errors. In particular, the kinetic energy is discretely bounded provided a symplectic time integration scheme is used, resulting in a discretely nonlinearly stable scheme.\(^7\) An alternative approach to ensure discrete nonlinear stability for incompressible Navier-Stokes simulations is to formulate the inviscid fluxes in “skew-symmetric” form.\(^3\) This form also ensures the discrete conservation of kinetic energy in the absence of viscosity, while enforcing discrete conservation of mass and momentum. This skew-symmetric formulation has been widely used for simulations of turbulent flows using spectral methods as well as non-dissipative central-difference schemes.

For the compressible Navier-Stokes equations the development of nonlinearly stable schemes is complicated by the fact that the nonlinear fluxes are rational functions of the conservative variables. The situation
is further complicated by the presence of shocks, though for the present discussion we consider only subsonic flows. Multiple avenues have been pursued for the development of nonlinearly stable schemes for compressible flows, which we discuss in the following paragraphs.

Total variation diminishing (TVD) MUSCL schemes based on approximate Riemann solvers have been widely used to ensure stability for compressible flow simulations. While these schemes have been used for DNS of compressible flows, the 2nd-order formulations typically used have too much numerical dissipation, and can result in poor predictions in the case of marginally resolved DNS or large eddy simulations (LES).

Non-dissipative central difference schemes have also been widely used for the simulations of turbulent compressible flows. Non-linear instabilities may be controlled using higher-order filtering schemes or hyper-viscosity methods, however these may introduce non-physical dissipation.\(^8\) Along the lines of approaches used for incompressible flows, skew-symmetric formulations have been developed for non-dissipative schemes which attempt to globally conserve either entropy or kinetic energy in the absence of viscosity or shocks.\(^4\)\(^-\)\(^6\),\(^8\)\(^,9\) However, such skew-symmetric schemes do not always ensure discrete conservation of both energy and entropy. Additionally, the effectiveness of such schemes may be limited to moderate Reynolds numbers.\(^8\) Furthermore, to the authors' knowledge, extensions to finite element formulations do not exist.

An alternative approach to develop nonlinearly stable central-difference schemes has been to use locally adapted stencils (i.e. limiters or WENO) schemes. A particularly promising approach uses a comparison technique to develop a WENO scheme for a summation-by-parts, simultaneous approximation term (SBP-SAT) finite-difference method, which is discretely entropy stable.\(^10\) Unfortunately, extension of this approach to general unstructured, finite element formulations is unclear.

For higher-order finite-element discretizations of compressible flows, nonlinear instabilities have been suppressed by using more accurate quadrature rules for the integration of the nonlinear fluxes.\(^11\) Such “polynomial-dealiasing” attempts to directly address the inexact projection of the nonlinear fluxes using collocated quadrature rules, and has been successfully applied with higher-order discontinuous Galerkin discretizations at moderate Reynolds numbers.\(^11\)\(^,12\) However, as we have previously demonstrated, this approach is not general.\(^13\)

Provably entropy-stable formulations for compressible flows have been developed based on stabilized continuous and discontinuous Galerkin finite-element methods discretized using entropy variables.\(^14\)\(^,15\) Despite their rigorous mathematical underpinning, finite-element formulations using entropy variables have been less widely used, in part, because they cannot be combined with explicit time-marching schemes and are viewed as too expensive. Additionally, low-order simulations have suggested that the benefits of the entropy variable formulations are largely theoretical, with little practical difference to conservative variables.\(^16\) In this work we demonstrate that for higher-order simulations of turbulent compressible flows, a non-linearly stable entropy variable formulation is necessary in order to maintain stability for marginally or under-resolved simulations.

We present a higher-order space-time discontinuous-Galerkin (DG) finite-element method for the simulation of turbulent compressible flow. Through numerical experiments we assess the nonlinear stability of different formulations using collocated and dealiased schemes with conservative or entropy variables. We demonstrate that with increasing Reynolds number and polynomial solution order, nonlinear stability is achieved only through a formulation consistent with a provably-stable formulation. Furthermore, we demonstrate that such a scheme may be implemented at reasonable cost.

For high Reynolds number flows, we demonstrate that the numerical stabilization in the DG scheme acts as an implicit subgrid model. For underresolved flows at low Mach number, upwind schemes can lead to a positively biased pressure-dilatation. In order to suppress this bias, a low-Mach variant of the approximate Riemann solver of Ismail and Roe\(^17\) is presented based on the ideas of low-Mach preconditioning.\(^18\)\(^,19\) The low-Mach modified flux successfully suppresses the upwind bias in the pressure-dilatation, while maintaining a form consistent with entropy stability. We further examine the effect of the choice of the Riemann solver on the implicit subgrid scale effects, in particular on the kinetic energy spectrum at high Reynolds number.

This paper begins with a brief description of the governing equations and our space-time DG discretization in Section II. In Section III we present the Taylor-Green vortex problem which is used as a model problem throughout. In Section IV we discuss aliasing instabilities for marginally resolved simulations at moderate Reynolds number. In Section V we discuss nonlinear stability for under-resolved simulations at high- and infinite Reynolds number. In Section VI we discuss the implicit subgrid-scale modeling in under-resolved high-Reynolds number flows and the effects of the choice of numerical flux. This section includes the presentation of our low-Mach modified Riemann solver. Finally, a summary and conclusions are presented.
in Section VII.

II. Numerical Method

The compressible Navier-Stokes equations are written in conservative form as:

\[ \mathbf{u}_t + \nabla \cdot (\mathbf{f}^I - \mathbf{f}^V) = 0 \]  

(1)

where \( \cdot \) denotes partial differentiation with respect to time. The conservative state vector is

\[ \mathbf{u} = \begin{bmatrix} \rho \\ \rho \mathbf{v} \\ \rho E \end{bmatrix}, \]

(2)

where \( \rho \) is the density, \( \mathbf{v} \) is the velocity, and \( E \) the total energy. The inviscid and viscous fluxes are given, respectively, by:

\[ \mathbf{f}^I = \begin{bmatrix} \rho \mathbf{v} \\ \rho \mathbf{v} \mathbf{v}^T + p \mathbf{I} \\ \rho \mathbf{v} H \end{bmatrix}, \quad \text{and} \quad \mathbf{f}^V = \begin{bmatrix} 0 \\ \tau \\ \tau \mathbf{v} + -\kappa T \nabla T \end{bmatrix}, \]

(3)

where \( p \) is the static pressure, \( H = E + \frac{\rho}{\gamma} \) is the total enthalpy, \( \tau \) the viscous stress tensor, \( \kappa T \) is the thermal conductivity, \( T = \rho / \rho R \) is the temperature, and \( R \) is the gas constant. The pressure is given by:

\[ p = (\gamma - 1) \left( \rho E - \frac{1}{2} \rho \mathbf{v}^2 \right), \]

(4)

where \( \gamma \) is the specific heat ratio. The viscous stress tensor, \( \tau \), is given by:

\[ \tau = \mu \left( \nabla \mathbf{v} + \nabla \mathbf{v}^T \right) - \lambda (\nabla \cdot \mathbf{v}) \mathbf{I} \]

(5)

where \( \mu \) is the viscosity, \( \lambda = \frac{2}{3} \mu \) is the bulk viscosity.

Applying a change of variables \( \mathbf{u} = \mathbf{u}(\mathbf{v}) \), where \( \mathbf{v} \) are the entropy variables:

\[ \mathbf{v} = \begin{bmatrix} -\frac{s}{2-1} + \frac{\gamma+1}{2} + \frac{\rho E}{\rho} \\ \frac{\rho V}{\rho} \\ -\frac{e}{\rho} \end{bmatrix} \]

(6)

we rewrite the Navier-Stokes equations as:

\[ A_0 \mathbf{v}_t + \mathbf{A} \nabla \mathbf{v} - \nabla \cdot (\mathbf{K} \nabla \mathbf{v}) = 0 \]

(7)

with symmetric \( A_0 = \mathbf{u} \cdot \nabla \), \( \mathbf{A} = \mathbf{f}^I_\mathbf{u} A_0 = \mathbf{f}^I_\mathbf{v} \) and \( \mathbf{K} = \mathbf{f}^V_\mathbf{v} A_0 = \mathbf{f}^V_\mathbf{v} \).

We proceed to discretize (1) or (7) as follows. The spatial domain, \( \Omega \), is partitioned into non-overlapping hexahedral elements, \( \kappa \), while the time is partitioned into time intervals (time-slabs), \( I^n = [t^n, t^{n+1}] \). Define \( \mathcal{V}_h = \{ \mathbf{w}, \mathbf{w} \} \in [\mathcal{P}(\kappa \times I)]^5 \}, \) the space-time finite-element space consisting of piece-wise polynomial functions in both space and time on each element. The weak form of the governing equations is given by:

\[ \sum_{\kappa} \left\{ \int_{t^n}^{t^{n+1}} - (\mathbf{w}_t \mathbf{u} + \mathbf{w}_x (\mathbf{f}^I_\mathbf{u} - \mathbf{f}^V_\mathbf{v})) + \int_{t^n}^{t^{n+1}} \int_{\partial \kappa} \mathbf{w} (\mathbf{f}^I_\mathbf{v} n_i - \mathbf{f}^V_\mathbf{v} n_i) + \int_{t^n}^{t^{n+1}} \mathbf{w} (t_{n+1}^{n+1} u(t_{n+1}^{n+1}) - w(t_{n+1}^{n+1}) u(t_{n+1}^{n+1})) = 0 \right\} \]

(8)

where the second and third integrals arise due to the spatial and temporal discontinuity, respectively, of the basis functions. \( \mathbf{f}^I_\mathbf{v} n_i \) and \( \mathbf{f}^V_\mathbf{v} n_i \) denote single valued numerical flux functions approximating, respectively, the inviscid and viscous fluxes at the spatial boundaries of the elements. In this work, the inviscid flux (unless otherwise noted) is computed using the Roe flux,\(^{17,20}\) while the viscous flux is computed using the method of Bassi and Rebay.\(^{21}\) When discretizing using conservative variables, we seek a solution \( \mathbf{u} \in \mathcal{V}_h \) and require (8) to be satisfied for all \( \mathbf{w} \in \mathcal{V}_h \). Similarly, when discretizing using the entropy variables, we seek
a solution $u(v)$, $v \in V_h$ which satisfies the weak form (8) for all $w \in V_h$. The space, $V_h$ is spanned by the tensor product of 1D nodal Lagrange basis functions defined at the Gauss-Legendre points.

Under exact integration and conditions on the choice of numerical flux function, the space-time DG discretization using entropy variables satisfies a discrete Clausius-Duhem inequality ensuring discrete satisfaction of the second law of thermodynamics. In particular, this allows for the proof of a nonlinear entropy-stability for the numerical scheme. In practice, integrals appearing in (8) are approximated with numerical quadrature rules. We consider a collocated scheme (where solution points are used as quadrature points), and polynomial dealiasing rules using more accurate quadrature. The choice of quadrature rules is discussed further in Section IV. In Section V, we consider both a conservative variable formulation and an entropy variable formulation consistent with the nonlinear stability proof.

With the choice of basis and an appropriate quadrature rule to perform numerical integration, (8) gives a system of nonlinear equations which must solved for each time-slab. In this work we use a preconditioned Jacobian-free Newton-Krylov solver we have previously presented.

### III. Taylor-Green Vortex Problem

The Taylor-Green vortex evolution is used as a model problem for turbulent flow as it involves only periodic boundary conditions, no forcing and a simple initial condition. The flow is solved on an isotropic domain, which spans $[0, 2\pi L]$ in each coordinate direction. The initial conditions are given by:

\begin{align*}
    u &= V_0 \sin(x/L) \cos(y/L) \cos(z/L) \\
    v &= -V_0 \cos(x/L) \sin(y/L) \cos(z/L) \\
    w &= 0 \\
    p &= \rho_0 V_0^2 \left[ \frac{1}{\gamma M_0^2} + \frac{1}{16} (\cos(2x/L) + \cos(2y/L)) (\cos(2z/L) + 2) \right]
\end{align*}

where $u$, $v$ and $w$ are the components of the velocity in the $x$-, $y$- and $z$-directions, $p$ is the pressure and $\rho$ is the density. The Taylor-Green vortex flow is simulated using the compressible Navier-Stokes equations at Mach number $M_0 = 0.1$. The flow is initialized to be isothermal ($\frac{p}{\rho} = \frac{p_0}{\rho_0} = R T_0$). Simulations are performed at a variety of Reynolds numbers $Re = \frac{\rho_0 V_0 L}{\mu}$.

Starting from the simple initial condition, the flow becomes turbulent through repeated vortex stretching leading to progressively smaller eddies, which are then dissipated to heat through the action of molecular viscosity. With increasing Reynolds number, progressively smaller structures appear. Figure 1 shows the iso-contours of vorticity at the instant of peak dissipation from a 16th-order solution at $Re = 1,600$. 

![Figure 1. Iso-contours of vorticity magnitude at the instant of peak dissipation for the Taylor-Green vortex evolution at $M = 0.1$, $Re = 1,600$, computed using conservative variables, with $256^3$ degrees of freedom.](image)
For each simulation the temporal evolution of the kinetic energy

\[ E_k = \frac{1}{\Omega} \oint_{\Omega} \frac{1}{2} \rho \mathbf{V} \cdot \mathbf{V} d\Omega \]  

(13)
is monitored. The evolution of the kinetic energy dissipation rate \( \epsilon = -dE_k/dt \) was computed based on the data at the space-time quadrature points. We assess the quality of our numerical solutions by computing individual terms in the kinetic energy evolution equation. For compressible flow, the kinetic energy dissipation rate is given by the sum of three contributions \( \epsilon = \epsilon_1 + \epsilon_2 + \epsilon_3 = -dE_k/dt \):

\[ \epsilon_1 = \frac{1}{\Omega} \oint_{\Omega} 2\mu \mathbf{S} : \mathbf{S} d\Omega \]  

(14)

\[ \epsilon_2 = \frac{1}{\Omega} \oint_{\Omega} \lambda (\nabla \cdot \mathbf{V})^2 d\Omega \]  

(15)

\[ \epsilon_3 = -\frac{1}{\Omega} \oint_{\Omega} p (\nabla \cdot \mathbf{V}) d\Omega \]  

(16)

where \( \mathbf{S} = \frac{1}{2}(\nabla \mathbf{V} + \nabla \mathbf{V}^T) \) is the strain rate tensor. Since the flow is nearly incompressible, we expect the dissipation due to the bulk viscosity, \( \epsilon_2 \), and the pressure-dilatation term, \( \epsilon_3 \), to be small. The kinetic energy dissipation rate is then approximately equal to \( \epsilon \approx \epsilon_1 \). However, for the compressible simulation this does not hold exactly.\(^{22}\)

### IV. Aliasing Errors

We present initial validation of our numerical method by performing direct numerical simulation (DNS) of the Taylor-Green vortex problem at \( Re = 160 \). The Reynolds number for this problem is sufficiently small that DNS may be performed with a relatively small number of degrees of freedom. We perform a mesh refinement study using our space-time DG scheme with 2nd-, 4th-, 8th-, 16th-order discretizations in space and 4th-order discretization in time. We use a formulation with conservative variables using both a collocated scheme and a polynomial dealiasing rule using \( 2N \) points, where \( N \) is the solution order.

We perform a mesh refinement study using 48, 64, 96 and 128 degrees of freedom in each coordinate direction. The integral of the error in the dissipation rate as a function of the mesh size, \( h = 1/DOF^{1/3} \), is presented in Figure 2. At low-order, (i.e. 2nd- and 4th-order) the numerical dissipation of the DG scheme plays a more significant role than the quadrature errors, so that the results for collocated and dealiaised schemes are essentially identical. Using the 16th-order scheme, at this low Reynolds number the errors introduced through the use of collocated quadrature do not induce nonlinear instabilities. While not leading to nonlinear instabilities, the collocated scheme generates errors in the projection of the fluxes onto the polynomial space and large errors in the computed kinetic energy dissipation rate. In particular, using 48 DOF in each coordinate direction, the inexact quadrature of the collocated 16th-order scheme results in an error in the computed kinetic energy dissipation rate which is an order of magnitude larger than the corresponding dealiased solution. With mesh refinement, both the solution and fluxes are well approximated in the polynomial space and thus the quadrature errors due to the collocated rule become less important. Both collocated and dealiased schemes demonstrate optimal error convergence rates, except at 16th-order where suboptimal convergence may be due to not yet having reached the asymptotic limit.

Next, we consider the Taylor-Green vortex problem at a larger Reynolds number of \( Re = 1,600 \). Figure 3 shows the corresponding temporal evolution of the kinetic energy dissipation rate \( -dE_k/dt \) computed with dealiased solutions using 256 degrees of freedom in each coordinate direction using 2nd- 4th-, 8th-, and 16th-order DG schemes. The corresponding meshes have, respectively, 128, 64, 32 and 16 elements in each coordinate direction. Reference data computed from an incompressible simulation using a spectral code on a \( 512^3 \) grid\(^{23}\) is also presented. For the 2nd-order scheme, there remains significant numerical dissipation at this resolution and the point of peak dissipation is poorly captured. With increasing solution order, the results relative to the spectral data are significantly improved. The dissipation rate for 4th-, 8th- and 16th-order schemes are nearly identical to the spectral data. However, zooming in at the point of peak dissipation, Figure 3(b) shows that 8th- and 16th-order schemes give somewhat better matching with the reference spectral data.

As we have seen with the lower Reynolds number case, higher-order schemes provide significant benefit for well resolved simulations. However, more often than not, turbulent flow simulations are marginally or under-
Figure 2. Error in kinetic energy evolution for Taylor-Green vortex problem at $M = 0.1$, $Re = 160$ computed using conservative variables. (solid line corresponds to the collocated scheme, dashed line corresponds to polynomial dealiasing using $2N$ points)

Figure 3. Taylor-Green vortex problem at $M = 0.1$, $Re = 1,600$, computed with conservative variables using $256^3$ degrees of freedom.
resolved. In these under-resolved simulations, higher-order schemes may provide benefit over lower-order schemes, however, due to the low numerical dissipation present in these schemes, nonlinear instabilities may arise if proper care is not taken. Gassner et al.\textsuperscript{12} demonstrated significant benefits in terms of computational efficiency using a explicit high-order DG scheme for computing a marginally resolved simulation of the Taylor-Green vortex at $Re = 1,600$. In the following, we duplicate simulations of Gassner et al.\textsuperscript{12} and our previous simulations\textsuperscript{13} using our space-time DG scheme. Figure 4(a) shows the evolution of the kinetic energy dissipation rate computed using a collocated DG scheme with a 4th-order temporal discretization and 2nd-, 4th-, 8th- and 16th-order spatial discretization using 64 DOF in each coordinate direction. At 2nd- and 4th-order, the numerical dissipation present in the DG scheme results in significant kinetic energy dissipation and we show very poor agreement with the reference spectral data. However, the numerical dissipation is sufficient to maintain stability of the DG scheme. At 8th- and 16th-order, the aliasing errors present in the collocated scheme cause nonlinear instabilities which eventually lead to catastrophic failure of the simulations.

![Figure 4. Taylor-Green vortex problem at $M = 0.1$, $Re = 1,600$ computed using meshes with 64\textsuperscript{3} degrees of freedom using a) collocated quadrature, and b) polynomial de-aliasing with 2N points](image)

The corresponding evolution of the kinetic energy dissipation rate using a polynomial dealiasing with 2N quadrature points is shown in Figure 4(b). The use of a more accurate quadrature rule reduces the aliasing errors such that 8th- and 16th-order simulations do not suffer from the nonlinear instabilities seen with the collocated scheme. Even for this marginally resolved simulation, there are clear benefits in using the higher-order scheme as the 8th- and 16th-order schemes compute a kinetic energy dissipation rate which matches well with the reference spectral DNS despite using an order of magnitude fewer degrees of freedom in each coordinate direction.

V. Entropy Stability

In the previous section, we have shown that nonlinear instabilities may be suppressed using polynomial dealiasing for marginally resolved higher-order simulations of turbulent flows at moderate Reynolds numbers. Next, we consider simulation of turbulent flows at even higher Reynolds numbers. In Figure 5(a) we show the evolution of the kinetic energy dissipation rate for a simulation of the Taylor-Green vortex problem at a Reynolds number of $Re = 16,000$ using our space-time DG scheme using a polynomial dealiasing rule with 2N points, discretized with conservative variables. A 4th-order temporal and 8th- and 16th-order spatial discretizations are used with 48 DOF in each coordinate direction. Using a polynomial dealiasing rule with 2N points is insufficient to suppress nonlinear instabilities at high order, leading to catastrophic failure of the simulations. Furthermore, using a dealiasing rule with 3N or 4N points (not shown) does not improve the situation. Clearly, at high-order, the dissipation in the DG scheme is insufficient to ensure the suppression of the nonlinear instabilities for under-resolved high Reynolds number turbulent simulations using conservative variables.

The corresponding kinetic energy dissipation rate, computed using the space-time DG scheme discretized...
Figure 5. Kinetic energy dissipation rate for the Taylor-Green vortex problem at $M = 0.1$, $Re = 16,000$ computed using meshes with $48^3$ degrees of freedom using polynomial dealiasing with $2N$ points discretized with a) conservative variables and b) entropy variables.

Figure 6. Entropy integrated over the domain for the Taylor-Green vortex problem at $M = 0.1$, $Re = 16,000$ computed using 8th-order spatial discretization with 48 degrees of freedom in each coordinate direction using polynomial dealiasing with $2N$ points.
with entropy variables and using a polynomial dealiasing rule with 2N points, is shown in Figure 5(b). Using the entropy variables, the scheme is consistent with the nonlinear entropy stability proof and the 8th- and 16th-order simulations do not exhibit nonlinear instabilities present when using conservative variables. The time history of entropy integrated over the domain using both conservative and entropy variables at 8th-order is presented in Figure 6. The simulations using conservative variables demonstrate non-physical entropy decay at the point of failure. On the other hand, simulations using entropy variables show monotonic entropy generation throughout the simulation.

VI. Effect of Numerical Flux

We have demonstrated the nonlinear stability properties of our space-time DG scheme discretized using entropy variables and polynomial dealiasing to perform under-resolved turbulent simulations at high Reynolds numbers. In this section we further demonstrate the nonlinear stability properties of our scheme by simulating the Taylor-Green vortex problem through infinite Reynolds number. We examine the sub-grid scale modeling effects of the DG scheme and examine the effect of the choice of numerical flux function.

First, we highlight the nonlinear stability of our space-time DG scheme by performing simulation of the Taylor-Green vortex problem at $Re = 160,000$ using a 4th-order temporal and 8th-order spatial discretization on a sequence of meshes using 32, 48, 64, 96 and 128 DOF in each spatial direction. The kinetic energy dissipation rate for each simulation is plotted in Figure 7. With increasing mesh resolution, increasing frequency content is better represented on the discrete mesh and thus the peak of the computed kinetic energy dissipation rate increases. The numerical dissipation present in the DG scheme acts as a subgrid scale model which reduces the effective Reynolds number.

![Figure 7. Kinetic energy dissipation rate for the Taylor-Green vortex problem at $M = 0.1$, $Re = 160,000$ computed using 8th-order spatial discretization.](image)

To investigate the subgrid scale effects of the numerical flux function, we perform simulations of the Taylor-Green vortex problem at moderate and infinite Reynolds number. (We note that higher-order simulations are only possible because of the nonlinear stability of the dealiased entropy variable formulation). We consider four different numerical flux functions: the Lax-Friedrichs flux, the Roe flux, the flux of Ismail and Roe, and the discrete symmetric mean-value flux. The original flux function of Roe does not ensure discrete entropy production at an interface, and as such is not consistent with the nonlinear stability proof. The scheme of Ismail and Roe was developed to address this, and presented as a cheaper alternative to the discrete symmetric mean-value flux which requires numerical integration of the flux Jacobian between the two interface states. For our higher-order DG scheme, computation of numerical fluxes comprises a much smaller fraction of the total computational time than in a second-order finite volume formulations. Thus, the added cost of additional flux Jacobian evaluations required for the discrete symmetric mean-value flux is readily absorbed.

The evolution of the Taylor-Green vortex is simulated at $Re = 1,600$ using the 8th-order DG scheme
with the four different numerical flux functions. The simulations are performed using 64 degrees of freedom in each coordinate direction. The kinetic energy dissipation rate, the dissipation due to viscous strain and the pressure-dilatation term are presented in Figure 8. The last three fluxes have identical linearized forms, thus we expect little difference between them for this subsonic flow which only involve small jumps across element interfaces. This is borne out by the simulations results. The Lax-Friedrichs flux shows significantly different behavior from the other three fluxes. In particular, using the Lax-Friedrichs flux results in larger viscous dissipation due to strain. We note that all fluxes exhibit a positively biased dissipation due to the pressure-dilatation term as we have previously reported for our DG scheme.\textsuperscript{22}

![Figure 8. Kinetic energy dissipation for the Taylor-Green vortex problem at $M = 0.1$, $Re = 1,600$ computed using 8th-order spatial discretization. (Flux options are Lax-Friedrichs, Ismail and Roe,\textsuperscript{17} Barth,\textsuperscript{15} and Roe\textsuperscript{20})]

In order to attempt to address the positive bias in the pressure-dilatation term seen with all of the upwind schemes considered, we have modified the scheme of Ismail and Roe with a dissipation inspired by low-Mach preconditioning techniques.\textsuperscript{18,19} In compressible flows, the ratio of the magnitude of the convective eigenvalues to the acoustic eigenvalues scales with the Mach number. Low-Mach preconditioning methods apply a scaling of the temporal derivatives leading to a modified system of equations for which the magnitude of the eigenvalues corresponding to all characteristics are of the same order. This allows solutions to be obtained using explicit schemes with a time step restriction dictated by the convective speed as opposed to the acoustic speed. It has also been demonstrated that applying numerical dissipation corresponding to the preconditioned system results in more accurate numerical simulations in the limit of low Mach number. Inspired by this, we modify the dissipation term in the scheme of Ismail and Roe based on the ideas of Low-Mach preconditioning.

We consider the linearized Euler equations preconditioned using the method of Merkle et al.\textsuperscript{18,19}

\begin{equation}
\mathbf{u},t + \Gamma^{-1} \mathbf{f}^T_{,\mathbf{u}} \mathbf{u} = 0
\end{equation
where $f_{Ju}^I$ is the inviscid flux Jacobian, while $\Gamma$ is the preconditioning matrix given by:

$$
\Gamma = \begin{bmatrix}
\rho_p' & 0 & \rho_T' \\
\rho_p'V & 0 & \rho_p'H - (1 - \rho h_p) \\
\rho_p'VH(1 - \rho h_p) & \rho V^T & \rho V^T + \rho h_pT \\
\end{bmatrix} \begin{bmatrix} y \\ u \end{bmatrix}
$$

(18)

where $y = \begin{bmatrix} \rho & V^T & T \end{bmatrix}^T$ is the vector of primitive variables, $T$ is the temperature, while $\rho_p'$ and $\rho_T'$ are artificial fluid properties implicitly defined by two non-dimensional parameters $\beta$ and $\delta$:

$$
\beta = \frac{\rho h_p' + \rho_T'(1 - \rho h_p)}{\rho_p'H - (1 - \rho h_p)} \quad \delta = \frac{\rho_T'}{\rho_p'}.
$$

(19)

Here, $\beta$ scales as $M_0^2$ while $\delta = 1$. By setting $\beta = \delta = 1$ in (17), we recover the unpreconditioned linearized Euler equations. The preconditioned flux Jacobian $\Gamma^{-1} f_{Ju}^I n$ has convective eigenvalues $\lambda = v_n = \nabla \cdot n$ and acoustic eigenvalues $\lambda = \frac{v_n(1+\beta)}{2} \pm c'$, where $c' = \sqrt{(v_n \frac{1+\beta}{2}) + \beta c^2}$ is the preconditioned speed of sound, while $c$ is the speed of sound.

In the scheme of Ismail and Roe, the numerical dissipation has the form $\frac{1}{2} R|\Lambda|R^T \Delta v$, where $R$ and $\Lambda$ correspond, respectively, to the right eigenvectors and eigenvalues of $f_{Ju}^I n$ (i.e. $f_{Ju}^I n = RAR^{-1}$). Recall also that $f_{\Delta v}^I = RAR^T$ and $R^T = u_v$. The numerical dissipation ensures entropy stability since $\frac{1}{2} \Delta v^T R|\Lambda|R^T \Delta v > 0$ by construction. We wish to modify the dissipation in the scheme of Ismail and Roe, while requiring that the modified numerical dissipation ensures entropy stability. To this end we propose a modified dissipation term of the form: $\frac{1}{2} R|\Lambda|R^T \Delta v$ where $R_p$ and $\Lambda_p$ now correspond, respectively, to the right eigenvectors and eigenvalues of the preconditioned system $\Gamma^{-1} f_{Ju}^I n$. Specifically, $R_p$ are scaled to ensure that $R_pR_p^T$ symmetrizes the preconditioned Euler equations (17). Specifically, $R_p R_p^T = u_{\nabla p}$ where $\nabla p$ are the entropy variables for the preconditioned equations, (17), while $u_{\nabla p}$ is given by:

$$
\begin{bmatrix}
\beta \rho & \beta \rho V^T & \beta \rho E \\
\beta \rho V & \beta \rho V^T + p I & \beta \rho V^T + p V \\
\beta \rho E & \beta \rho E V^T + p V^T & \beta \rho E V^T + p V^T + (1 - \beta)p|V|^2 \\
\end{bmatrix}
$$

(20)

We now proceed to apply this modified scheme to perform direct numerical simulations of the Taylor-Green vortex evolution. Once again we perform the simulation at $Re = 1600$ using an 8th-order DG scheme with 64 degrees of freedom in each coordinate direction. We also consider the “entropy-conservative” scheme of Ismail and Roe (i.e. without dissipation). The evolution of the kinetic energy dissipation rate, viscous dissipation and pressure-dilatation term are presented in Figure 9. As desired, the modified scheme does not exhibit a biased pressure-dilatation term. The kinetic energy dissipation rate and the viscous dissipation given by this scheme are similar to the flux of Ismail and Roe. The non-dissipative entropy-conservative scheme also shows no bias in the pressure-dilatation term. At this Reynolds number and resolution the dissipation is due primarily to the physical viscosity which is sufficient to ensure stability even in the absence of numerical dissipation.

The kinetic energy spectrum for the simulations using the Lax-Friedrichs, Ismail and Roe, and low-Mach modified Ismail and Roe flux functions at a nondimensional time of $T = 11.8$ is presented in Figure 10. At this low Reynolds number the flow is nearly resolved using 64 degrees of freedom in each coordinate direction and there is little difference between the kinetic energy spectra computed using the different flux options at the higher mesh resolutions. Even at coarser resolutions the difference between the schemes is not very significant. Finally, we note that the kinetic energy decay spectrum has the expected slope of $-\frac{5}{3}$ up to a wave number corresponding to about half of the largest representative wave number on the given mesh.

In order to distinguish further between the different flux functions, we perform simulation of the Taylor-Green vortex problem at $Re = 16000$. Again we use the DG scheme with 4th-order temporal and 8th-order spatial discretizations. The kinetic energy spectrum using the Lax-Friedrichs flux, the flux of Ismail and Roe and the modified variant of this flux are plotted in Figure 11. At this higher Reynolds number, differences appear between the Lax-Friedrichs flux and the other two flux functions considered. Specifically, the kinetic energy spectrum computed using the Lax-Friedrichs flux does not exhibit an exponential decay with
Figure 9. Kinetic energy dissipation for the Taylor-Green vortex problem at $M = 0.1, Re = 1,600$ computed using 8th-order spatial discretization. (Flux options are Lax-Friedrichs, Ismail and Roe, Ismail and Roe with low-Mach modified dissipation and Ismail and Roe no numerical dissipation)

Figure 10. Kinetic energy spectrum for the Taylor-Green vortex problem at $M = 0.1, Re = 1,600$ computed using 8th-order spatial discretization. (solid lines correspond to flux of Ismail and Roe, dashed line correspond to Lax-Friedrichs flux, dash-dot line corresponds to Low-Mach modified flux.)
constant slope over the large wavenumbers. There appears to be less energy content at lower wavenumbers, and correspondingly more energy at higher wavenumbers compared with the flux of Ismail and Roe and the low-Mach variant.

Finally, we consider the limit of infinite Reynolds number. In this case the dissipation in the simulation is entirely the result of the numerical dissipation. As before we use our space-time DG scheme with 4th-order temporal and 8th-order spatial discretizations on a sequence of meshes. Figure 12 presents the kinetic energy spectrum at \( T = 11.8 \). Once again, we see marked differences between the Lax-Friedrich flux and the other two flux functions considered. The kinetic energy spectrum computed using the Lax-Friedrichs flux is no longer monotonic, and there appears to be large accumulation of energy at the higher wave numbers. On the other hand, the flux of Ismail and Roe and the low-Mach variant show monotonic decay of kinetic energy with increasing wavenumber.

VII. Summary

We have presented a higher-order space-time discontinuous-Galerkin finite-element method. We have demonstrated the importance of formulating the DG scheme in a manner consistent with existing nonlinear entropy-stability theory. In particular, we have demonstrated that using entropy variables and a polynomial dealiasing rule ensures nonlinear stability for under-resolved turbulent simulations at high-order.

The effect of different choices of numerical flux function has been investigated. A low-Mach variant of the flux function of Ismail and Roe has been presented which suppresses the biased pressure-dilatation observed with other flux functions. The modified flux presented maintains the entropy-stability properties of the
(a) Lax-Friedrichs

(b) Ismail & Roe

(c) Low-Mach

Figure 12. Kinetic energy spectrum for the Taylor-Green vortex problem at $M = 0.1$, $Re = \infty$ computed using 8th-order spatial discretization.
Finally, we have demonstrated the sub-grid scale modeling effects of the different choices of flux function by performing simulations of the Taylor-Green vortex problem at very high and infinite Reynolds number. The different choices of numerical flux function result in significantly different kinetic energy spectra. In particular, the Lax-Friedrichs flux led to a non-monotonic kinetic energy spectrum in the limit of infinite Reynolds number. Future work is required in order to assess the effect of numerical dissipation in combination with explicit sub-grid scale modeling.

References


